

AVIONS PIERRE ROBIN

BP 38

DIJON

France

SERVICE

SERVICE BULLETIN N° : 51

Corresponding to Modification n° 1166.

OBJECT : Modification of blanking plates, on rear fuselage sides.

AIRCRAFT CONCERNED : HR 200/100 and HR 200/120, up to n° 26 inclusive,  
except n° : 20, 23, 24 and 25.

Date : April 22nd 1974

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Date of application : See below

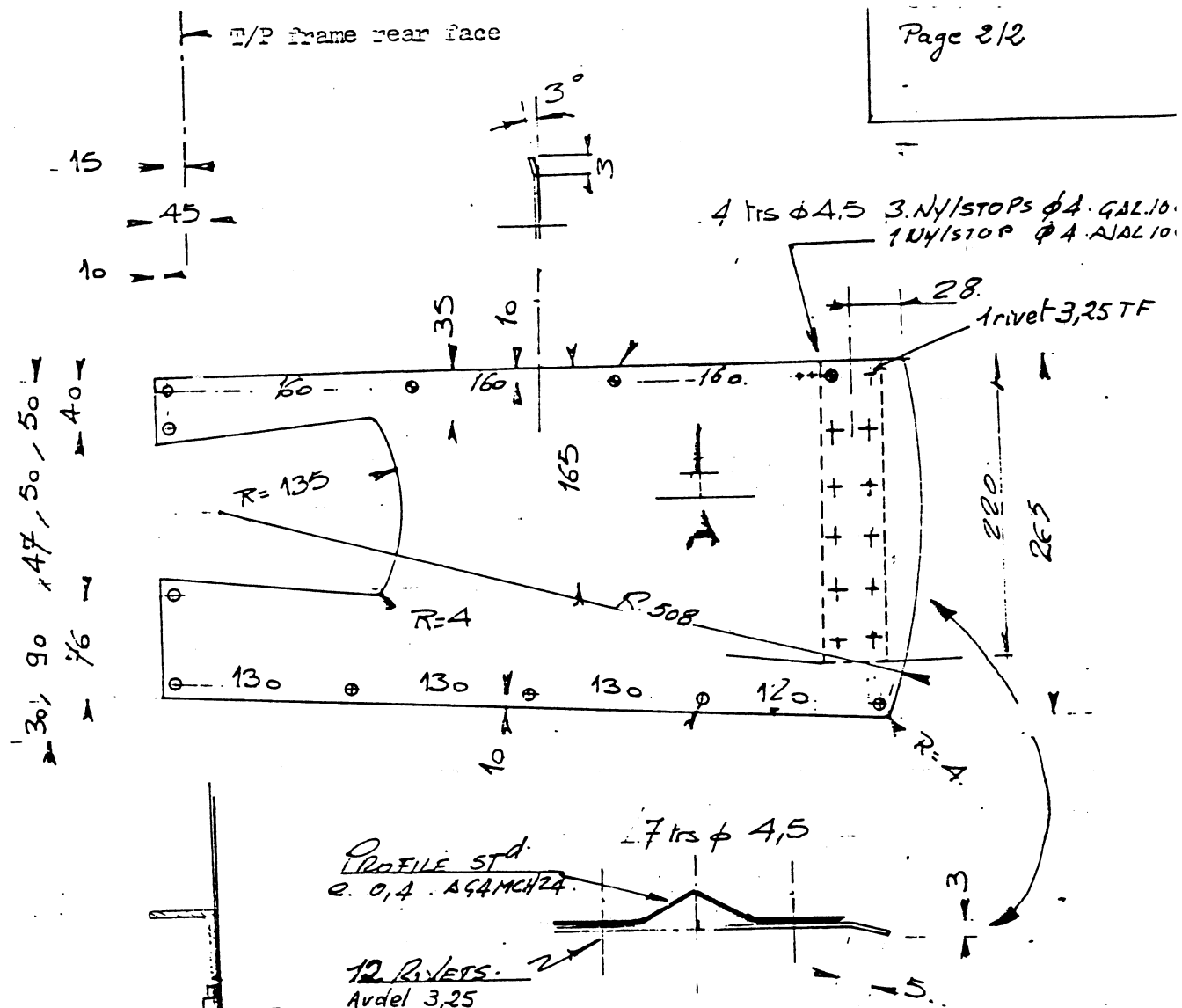
Approved by : S.G.A.C.

Upon receipt of this Service Bulletin, check to see if the fork-end fitting of the elevator tab control comes into contact with the end of the rear fuselage blanking plate, or if the said end of the plate is distorted. If this is the case, embody Modification n° 1166 (see drawing, page 2/2) as soon as possible. A ferrying flight can be carried out, after having straightened the plate rear end so as to prevent it from contacting the fork-end fitting.

In any case, Modification n° 1166 (which consists in rivetting a stiffener on the inside face of the plate, and in reducing the length of the latter) must be embodied, at the latest, within the next 20 flying hours and, in any case, before June 30th 1974.

Our Company, as well as our Service-Stations, are at your disposal, to carry out this modification.

IMPERATIVE



① LH side plate (drawing)

② RH side plate (symmetrical)

Mat: AG4 MCH24 ép: 0,5.

Blanking plate

Rear fuselage side

